Implosion Hydrodynamics for High Energy-Density Physics and Inertial Confinement Fusion



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What is nuclear fusion?



 $D + T \rightarrow \alpha$ (3.5MeV)+n(14.1MeV)

What could you do with a Terajoule? (~3g of DT)

 \rightarrow You can drive your car for 625,000 miles

 \rightarrow You can keep your furnace running for 8 years

 \rightarrow You can blow things up! 1TJ = 250 ton of TNT

Fusion fuel is produced from sea water



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Fusion doesn't come easy



A "hot plasma" at 100M ^oC is needed

- Probability for fusion reactions to occur is low at low temperatures because of Coulomb repulsion force.



 If the ions are sufficiently hot (i.e. large random velocity) then they can collide by overcoming Coulomb repulsion







SOLUTION:

\rightarrow Let the plasma do it itself!



THE α -PARTICLES HEAT THE PLASMA!

Under what conditions the plasma keeps itself hot?

Ignition condition P τ > 10 atm-sec

P= pressure in atmospheres
τ= confinement time in seconds

The plasma is too hot to be kept inside a solid container

SOLUTIONS:

1→Use a magnetic field to contain it. Since the plasma is made of charged particles (ions + electrons), a magnetic field should confine it. Magnetic Confinement Fusion P~atm, τ~sec, T~10⁸ °C

#2→Don't confine it! Or you can say that the plasma is confined by its own inertia Inertial Confinement Fusion P~Gigabar, τ~nanosecond, T~10⁸ °C The high pressures and temperatures required for inertial fusion can be achieved through laser-driven spherical implosions of a thin shell

Laser-driven spherical implosions can be used to:

→Achieve extreme states of matter of interest for inertial confinement fusion and/or general HEDP

- \rightarrow Achieve high temperatures (~ 10⁸⁻⁹ °C \rightarrow 10-100keV)
- \rightarrow Achieve high densities (~300-1000g/cc)
- \rightarrow Achieve high pressures (~10⁹⁻¹² atm \rightarrow Gbar-Tbar)
- \rightarrow Achieve high areal densities = ρR (~1-3g/cm²)

Laser-driven imploding capsules are mm-size shells with hundreds of μ m thick layers of cryogenic solid DT



LECTURE #1

Fundamentals of implosion hydrodynamics

We will use the conservation equations of gas-dynamics + ideal gas EOS to treat the DT plasma

$$\partial_t \rho + \partial_x (\rho u) = 0$$
 \leftarrow mass conservation

$$\partial_t (\rho u) + \partial_x (p + \rho u^2) = \text{body forces}(\text{ex}: \rho \vec{g})$$

 $\leftarrow \text{momentum conservation}$

$$\partial_t \varepsilon + \partial_x [v(\varepsilon + p) - \kappa \partial_x T] = \text{sources} + \text{sinks}$$

conservation

$$p = pressure = (n_e T_e + n_i T_i) = 2nT = (2/m_i)\rho_i T \qquad \stackrel{\leftarrow id}{equ}$$

Hideal gas equation of state

 $T_{e} = T_{i} = T = i/e \text{ temperature} \qquad n_{e} = n_{i} = n = i/e \text{ particle density}$ (for DT plasmas n_e=n_i) $\varepsilon = \frac{3}{2}p + \rho \frac{u^{2}}{2} \quad \leftarrow \text{Total energy per unit volume} \qquad \rho = \text{mass density} = n_{i}m_{i}$

 \mathbf{K} = plasma thermal conductivity

$$u =$$
velocity

The plasma thermal conductivity goes like a power law of T

$$n\frac{\partial T}{\partial t} = \frac{\partial}{\partial x} \kappa \frac{\partial T}{\partial x} \text{ Dimensional analysis } n\frac{T}{t} \sim \frac{\kappa T}{x^2} \implies \kappa \sim n\frac{x^2}{t}$$
$$x \Rightarrow \lambda_{mfp} \sim \upsilon_{th} \tau_{coll} \sim \frac{\upsilon_{th}}{\nu_{coll}} \qquad t \Rightarrow \tau_{coll} = \frac{1}{\nu_{coll}}$$
$$\kappa \sim n\frac{\upsilon_{th}^2}{\nu_{coll}} \qquad \upsilon_{th}^2 \sim \frac{T}{m_e} \qquad \nu_{coll} \sim \frac{n}{T^{3/2}} \implies \kappa \sim T^{5/2}$$

 v_{th} = thermal velocity v_{coll} = collision frequency τ_{coll} = collision time

Plasma thermal conductivity

$$\kappa \approx \kappa_0 T^{5/2}$$

Sound speed in an ideal DT gas/plasma

Adiabatic sound speed when the entropy is conserved along the fluid motion

$$C_s^{adiabatic} = C_s(\text{constant entropy}) = \sqrt{\frac{5}{3}\frac{p}{\rho}} = \sqrt{\frac{10}{3}\frac{T}{m_i}}$$

Isothermal sound speed when the temperature is constant along the fluid motion

$$C_s^{isothermal} = C_s$$
 (constant temperature) $= \sqrt{\frac{p}{\rho}} = \sqrt{\frac{2T}{m_i}}$





What is the pressure generated by the laser?

• Use the energy conservation equation

$$\partial_t \varepsilon + \partial_x \left[u(\varepsilon + p) - \kappa \partial_x T \right] = I \delta(x - x_c)$$

Energy deposited by the laser near critical W/cm³s

• Since the temperature gradients are small in the corona, the heat flux is small

$$\kappa \partial_x T(x \ge x_c) << \kappa \partial_x T(x \le x_c)$$

More accurately
$$\kappa \partial_x T(x \ge x_c) \approx \frac{1}{3} \kappa \partial_x T(x \le x_c)$$

Integrate around critical surface x_c

$$-\left[\kappa\partial_{x}T\right]_{x_{c}^{-}}^{x_{c}^{+}}=I \quad \Longrightarrow \quad \kappa\partial_{x}T(x_{c}^{-})=I$$

• Solving at steady state in the conduction zone (x<x_c) leads to

$$v(\varepsilon + p) \sim \kappa \partial_x T \quad \text{for } x \leq x_c$$

• At the sonic point (i.e. critical surface)

$$I = \left[u(\varepsilon + p)\right]_{x_c^-} = c_s \left(\frac{5}{2}p_c + \rho_c \frac{C_{s(isoT)}^2}{2}\right) \sim \frac{p_c^{3/2}}{\rho_c^{1/2}}$$

• The total pressure (static + dynamic) is the ablation pressure

$$P_{A} = [p + \rho u^{2}]_{x=x_{c}} = 2p_{c} \sim (I\rho_{c}^{1/2})^{2/3} \sim \left(\frac{I}{\lambda_{L}}\right)^{2/3}$$

• The laser-produced total (ablation) pressure on target is

$$P_{A} \approx 83Mbar \left(\frac{I_{15}}{\lambda_{L(\mu m)}/0.35}\right)^{2/3} \begin{vmatrix} I_{15} \text{ laser intensity} \\ \text{in } 10^{15} \text{ W/cm}^{2} \\ \lambda_{L(\mu m)} \text{ laser wavelengtl} \\ \text{in microns} \end{vmatrix}$$

What is the mass ablation rated induced by the laser?

 At steady state, the mass flow across the critical surface must equal the mass flow off the shell (i.e. the mass ablation rate m_a)

$$\dot{m}_a = \rho u = \rho_c u_c = \rho_c C_{s(isoT)}^{crit} = \rho_c \sqrt{\frac{p_c}{\rho_c}} = \sqrt{\rho_c p_c}$$



$$\dot{m}_a = 3.26 \bullet 10^5 \frac{I_{15}^{1/3}}{\lambda_{L(\mu m)}^{4/3}} g / cm^2 s$$

What is the entropy of an ideal gas/plasma?

• The entropy S is a property of a gas just like P, T and ρ

$$S = c_v \ln \left[\frac{p}{\rho^{5/3}} const \right] = c_v \ln[\alpha]$$

$$\alpha = const \frac{p}{\rho^{5/3}}$$

- \bullet We call α the "adiabat"
- The entropy/adiabat S/ α changes through dissipation or heat sources or sinks

$$\rho\left(\frac{\partial S}{\partial t} + \vec{u} \bullet \nabla S\right) = \frac{DS}{Dt} = \mu \frac{\left|\nabla \vec{u}\right|^2}{T} + \frac{\nabla \bullet \kappa \nabla T}{T} + \text{sources/sinks}$$

 In an ideal gas (no dissipation) and without sources and sinks, the entropy/adiabat is a constant of motion of each fluid element

$$\frac{DS}{Dt} = 0 \Longrightarrow S, \alpha = const \Longrightarrow p \sim \alpha \rho^{5/3}$$

A low adiabat (entropy) gas is easy to compress

• smaller $\alpha \rightarrow$ less work to compress from low to high density

$$W_{1\to 2} = -\int_{\rho_1}^{\rho_2} p dV \sim -\int_{\rho_1}^{\rho_2} \alpha \rho^{5/3} d\frac{M}{\rho} \sim \alpha M\left(\rho_2^{2/3} - \rho_1^{2/3}\right)$$

• smaller $\alpha \rightarrow$ higher density for the same pressure

$$\alpha \sim \frac{P}{\rho^{5/3}} \qquad \Rightarrow \qquad \rho \sim \left(\frac{P}{\alpha}\right)^{3/5}$$

 in HEDP, the constant in the definition of the adiabat comes from the normalization of the pressure with the Fermi pressure

$$\alpha \equiv \frac{p}{p_F} \implies \text{(for DT plasma)} \qquad \alpha_{DT} \equiv \frac{p(Mb)}{2.2\rho(g/cc)^{5/3}}$$

What is a shock?

 If a gas/plasma is rapidly compressed by a piston, the acoustic/compression waves launched by the piston overlap due to the always increasing sound speed of a compressed gas/plasma. This overlap causes a steepening of the hydro properties→SHOCK



The flow of mass, momentum and energy is conserved across the shock front \rightarrow Rankine-Hugoniot conditions

$$\rho_1 u_1 = \rho_2 u_2$$

$$p_1 + \rho_1 u_1^2 = p_2 + \rho_2 u_2^2$$

$$u_1(\varepsilon_1 + p_1) = u_2(\varepsilon_2 + p_2)^2$$

• For an ideal gas/plasma

 $\varepsilon = \frac{3}{2}p + \rho \frac{u^2}{2}$



• For example: assign, ρ_1 , p_1 and p_2 to find ρ_2 , u_2 , u_1 =-U_{shock} using the three R-H conditions

For a strong shock $(p_2 >> p_1)$ the R-H are simplified



In an ideal gas/plasma, the adiabat α is constant unless a shock is present that raises the adiabat



 Time required for the shock to reach the rear target surface (shock break-out time = t_{sb})

$$t_{sb} = \frac{\Delta_1}{u_{shock}} = \Delta_1 \sqrt{\frac{3\rho_1}{4p_2}} \propto \sqrt{\frac{1}{\alpha_2 \rho_1^{5/3}}}$$

• If the target is initially cryogenic solid DT at 18K, then

$$\rho_1 = 0.25g/cc \qquad \alpha = \frac{p_{foot}(Mbar)}{2.2} \qquad P_{foot} \approx 83Mbar \left(\frac{I_{15}^{foot}}{\lambda_{\mu m}/0.35}\right)^{2/3}$$

$$I \approx 4.3 \bullet 10^{12} \frac{W}{cm^2} \implies p_{foot} = 2.2Mbar \implies \alpha = 1$$
$$I \approx 1.2 \bullet 10^{13} \frac{W}{cm^2} \implies p_{foot} = 4.4Mbar \implies \alpha = 2$$

$$I \approx 2.2 \bullet 10^{13} \frac{W}{cm^2} \implies p_{foot} = 6.6Mbar \implies \alpha = 3$$

In order to accelerate a shell to high velocity without raising the adiabat, the pressure must be "slowly" increased after the first shock.



- After the foot of the laser pulse, the laser intensity must be raised starting at about $0.5t_{sb}$ and reach its peak at about t_{sb}
- Reaching I_{max} at t_{sb} prevents a rarefaction/decompression wave to propagate back from the rear target surface and decompress the target.

Most of the laser energy absorbed by the plasma goes into the kinetic and thermal energy of the expanding blow-off plasma rather than into kinetic energy of the imploding shell





$$M \frac{du}{dt} = -4\pi R^2 P_a$$

Shell Newton's law

 $\frac{dM}{dt} = -4\pi R^2 \dot{m}_a$

Shell mass decreases due to ablation

 $P_a = \dot{m}_a u_{exhaust}$

Ablation pressure = Abl. Rate X exhaust vel.

The rocket model

Integrating the rocket equations yields the shell velocity, the shell mass and the hydro efficiency that depends on the ablated mass.

• Assume that driver (i.e. P_a) is on till the shell is about $\frac{1}{2}$ initial radius



Efficiency is maximum for $M_{final} \sim 0.2 M_{initial}$. This is not a good operating point for a DT ablator (Direct Drive) but works for a non-DT ablator (Indirect Drive) For Direct Drive $\eta_h \sim 8-10\%$.

Homework problem

- Design a direct-drive laser pulse for a cryogenic DT shell with inner radius R₁=1.35mm and outer radius R₀=1.7mm. Plot the laser power in Terawatts (10¹²W/cm²) versus time in nanoseconds (10⁻⁹ s).
- The laser always shines on the outer surface at r=R₀
- The total pulse energy is 1.5MJ
- The initial DT density is 0.25g/cc.
- The maximum laser intensity is 10¹⁵ W/cm²

Homework problem (continue)

- The foot of the laser pulse must set the shell on an adiabat α =3
- Calculate the shock break-out time t_{sb}
- Use a cubic power law in time to raise the intensity from 0.5t_{sb} to t_{sb}
- For a UV laser with λ_L =0.35µm, estimate the ablation pressure at I_{max}, the fraction of ablated mass and the hydro-efficiency
- Assuming 60% of laser energy absorption, estimate the final shell implosion velocity